

Mtg 28: Mon, 30 Oct 06 + 5 min (28-1)  
(Lect 4)

- Plan: - Eigenvalue pb: Rigid-body modes,  
mechanisms
- Beams: codes, distributed load,  
superposition.

### Eval pbs:

Appl (previously seen):

- Vibrations (vib. freq. & mode shapes)  
evals.

- Control: stability.

- FEAD: Rigid-body modes,  
mechanisms  
(stability)

- IEA: math

A

$n \times n$

, eval pb:

$$\underline{A} \underline{x} = \lambda \underline{x}$$

Standard eval pb.

Pam:  
Philip }

Eq. of motion

(28-2)

$$\underline{M} \ddot{\underline{d}} + \underline{C} \dot{\underline{d}} + \underline{K} \underline{d} = \underline{F}(t)$$

$n \times n \quad n \times 1 \quad n \times n \quad n \times 1 \quad \curvearrowright \quad n \times n \quad n \times 1 \quad n \times 1$

$\underline{M}$  = mass matrix (1)

$\underline{C}$  = damping mat.

$\underline{K}$  = stiffness mat.

$\underline{d}(t)$  = disp. dof's.

gen. disp (include rot's  
for beams)

Philip :  $\det(\lambda \underline{M} - \underline{K}) = 0$   
(w/ notes)

Russell :  $\det(\lambda \underline{I} - \underline{A}) = 0$

$$\underline{K} \underline{x} = \lambda \underline{M} \underline{x} \quad (2)$$

Generalized eval. pb.

Standard. eval. pb :  
for  $\underline{A}$

$$\underline{A} \underline{x} = \lambda \underbrace{\underline{I} \underline{x}}_{\underline{x}} \quad (3)$$

Shawn: Solving homog. diff. eq. (28-3)

( $\underline{F} = \underline{0} \Rightarrow$  Free vib. pb.)

+ undamped: ignore damping term

$\underline{C} = \underline{0}$

Undamped free vib.:

$$\boxed{\underline{M} \ddot{\underline{d}} + \underline{K} \dot{\underline{d}} = \underline{0}} \quad (1)$$

Goal: Try to find a basis of vectors to uncouple the syst. of coupled diff. eqs. (eq. of motion) Eq. (1) on p. 28-2.  
(matrix)

Method to find such basis: Eval pb Eq. (2) on p. 28-7, which came from Eq. (1) on p. 28-3 (undamped free vib. pb.)

How? Assume  $\dot{\underline{d}}(t)$  periodic in  $t$   
(actually sinusoidal)

$$\ddot{\underline{d}}(t) = (A \cos \omega t + B \sin \omega t) \underline{\phi}$$

$\xrightarrow[n \times 1]{\quad}$

rel.  $\dot{\underline{F}_0}$   
eval.

(eigenvect)  
 $\xrightarrow[n \times 1]{\quad}$

Sam: Just plug into Eq. (1), p. 28.3:

$$\ddot{\underline{d}}(t) = -\omega^2 \underbrace{(A \cos \omega t + B \sin \omega t)}_{\ddot{\underline{d}}} \underline{\phi}$$

(Philip: Plug in init. value &  $\frac{d}{dt} \uparrow$   
Not yet!)

$$\Rightarrow \boxed{\ddot{\underline{d}} = -\omega^2 \underline{d}}$$

$$\text{Eq. (1), p. 28.3: } -\omega^2 \underline{M} \underline{d} + \underline{K} \underline{d} = 0$$

$$\Rightarrow \underline{K} \underline{d} = \omega^2 \underline{M} \underline{d}$$

or

$$\boxed{\underline{K} \underline{\phi} = \omega^2 \underline{M} \underline{\phi}}$$

$\xrightarrow{x} \quad \uparrow \lambda \quad \xrightarrow{x}$

Eq. (2),  
p. 28.2.

Mtg 29: Wed, 1 Nov 06 + 5 min  
(Lect 5)

L29-1

## Stability of Trusses:

\* Standard eval pb of  $\underline{K}$ :

$$\underline{K} \underline{\zeta} = \lambda \underline{\zeta} = \lambda \underline{I} \underline{\zeta}$$

\* Find zero evals:

$\underline{K}$  = global stiff. mat. before  
acct. for any ess. b.c.

Two types of zero evals:

- { 1) rigid-body modes
- 2) mechanisms in trusses

→ zero-energy modes.

Type 1 zero evals: Rigid-body modes

1.1. 1-D struct:

1 transl. (1 zero eval.)

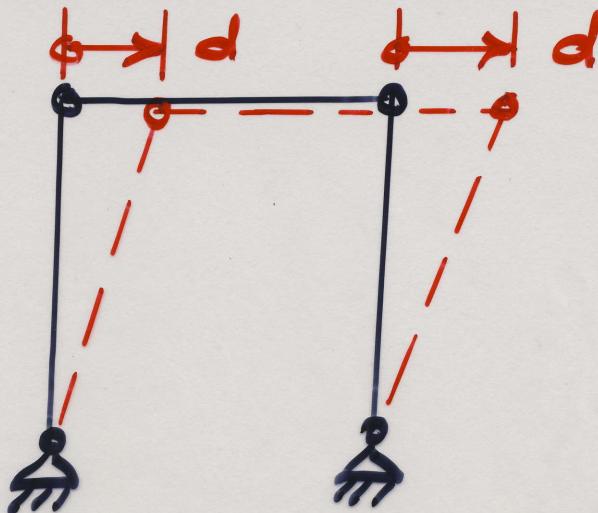
1.2. 2-D struct:

2 transl. + 1 rot.  
(3 zero eval.)

1.3. 3-D struct : 3 transl. + 3 rot. L<sup>29-2</sup>  
C zero evals.

Type 2 zero evals : Mechanisms

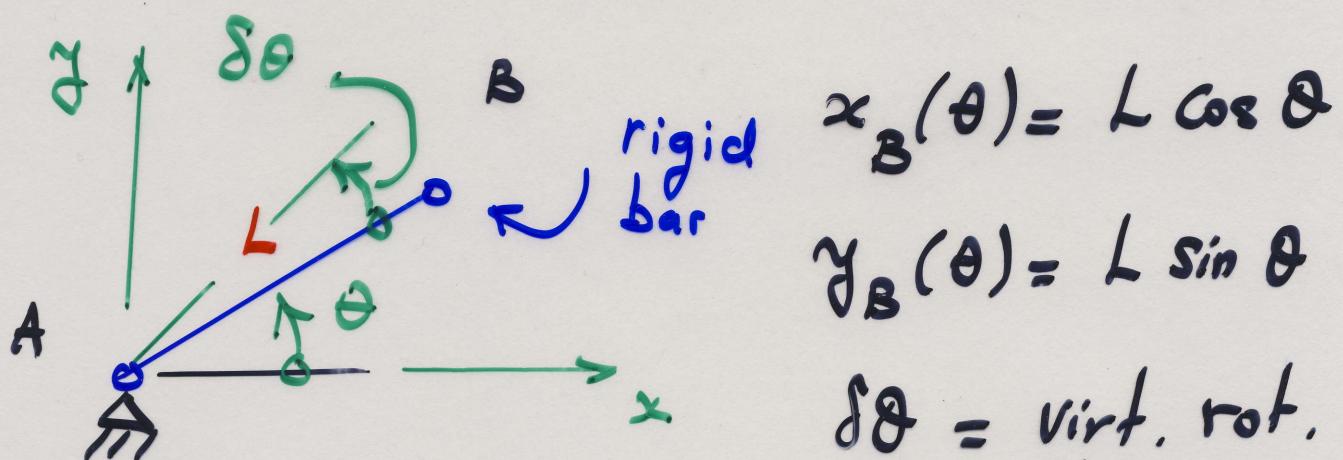
HW10:



mechanism w/o  
structural defm,  
i.e., w/o stored  
strain energy

Rem:  $\rightarrow$  zero-energy modes

Virtual disp (virt. velocity)



Q: What are the disp.  
at B imparted by  $\delta\theta$ ? (infinitesimally small)

$$\frac{\delta x_B}{\delta \theta} = \frac{d x_B}{d \theta} = -L \sin \theta$$

(29-3)

$$\Rightarrow \boxed{\delta x_B = \frac{d x_B}{d \theta} \delta \theta}$$

$$= (-L \sin \theta) \delta \theta$$

Similarly,  $\delta y_B = \frac{d y_B}{d \theta} \delta \theta$

$$= (L \cos \theta) \delta \theta$$

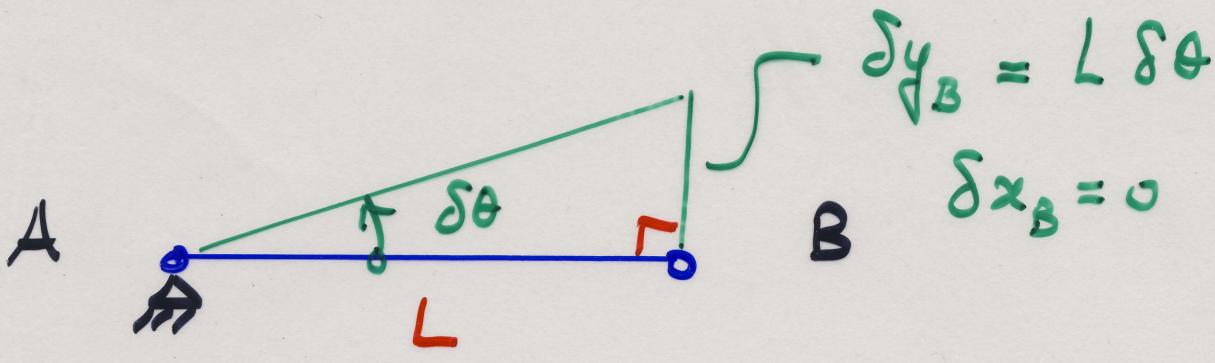
Virt disp comp.:  $\delta x_B(\theta), \delta y_B(\theta)$

Now consider an initially horiz bar:

$\boxed{\theta = 0}$ : What are the virt. disp now?

$$\delta x_B(\theta=0) = 0$$

$$\delta y_B(\theta=0) = L \delta \theta$$



≡

Evects corresp. to zero crals (99-4)  
 Come out from eig not pure  
 ( i.e., one cannot say which crests  
 Corresp. to rigid-body modes,  
 and which evects corresp. to  
 mechanisms ).

Consider :  $\underline{K} \underline{v} = \lambda \underline{v}$  ( for HW10  
 ex , p. 29-2 )

Let  $\underline{v}_1, \underline{v}_2, \underline{v}_3, \underline{v}_4$   
 $\underbrace{\underline{v}_1, \underline{v}_2, \underline{v}_3}_{\text{pure rigid body modes}}, \underbrace{\underline{v}_4}_{\text{mechanism}}$   
 ( plotted on p. 29-2 )

be evects corresp. to the 4 zero crals.

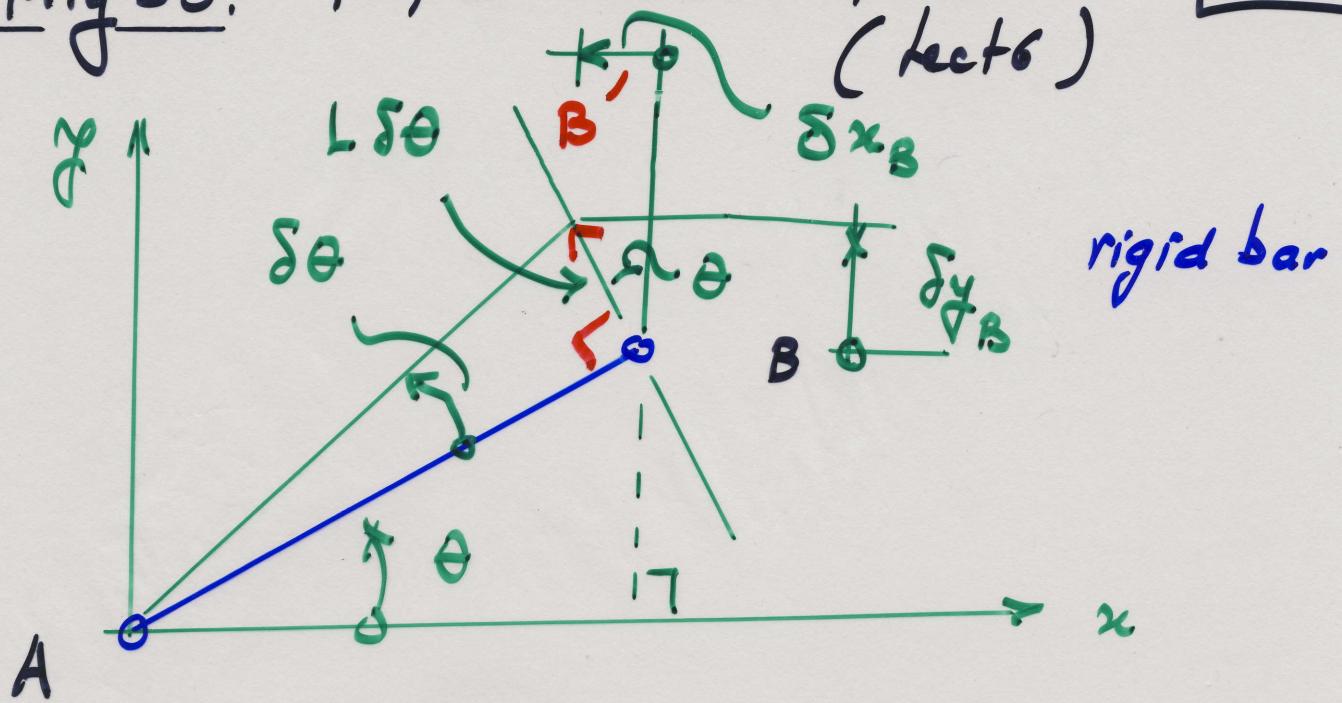
$$\underline{K} \underline{v}_i = 0. \underline{v}_i \quad i = 1, \dots, 4 \\ = 0$$

Now consider :  $\underline{w} = \sum_{i=1}^4 c_i \underline{v}_i$   
 for any  $c_i$ 's

$$\text{Amruta : } \underline{K} \underline{W} = \sum_i c_i (\underbrace{\underline{K} \underline{v}_i}_{\text{all}}) \quad \underline{L^{29-5}}$$
$$= \underline{0}$$

$\Rightarrow$   $\underline{W}$  is an erect comp. to  
a zero eval.

Mtg 30: Fri, 3 Nov 06 + 5 min L30-1



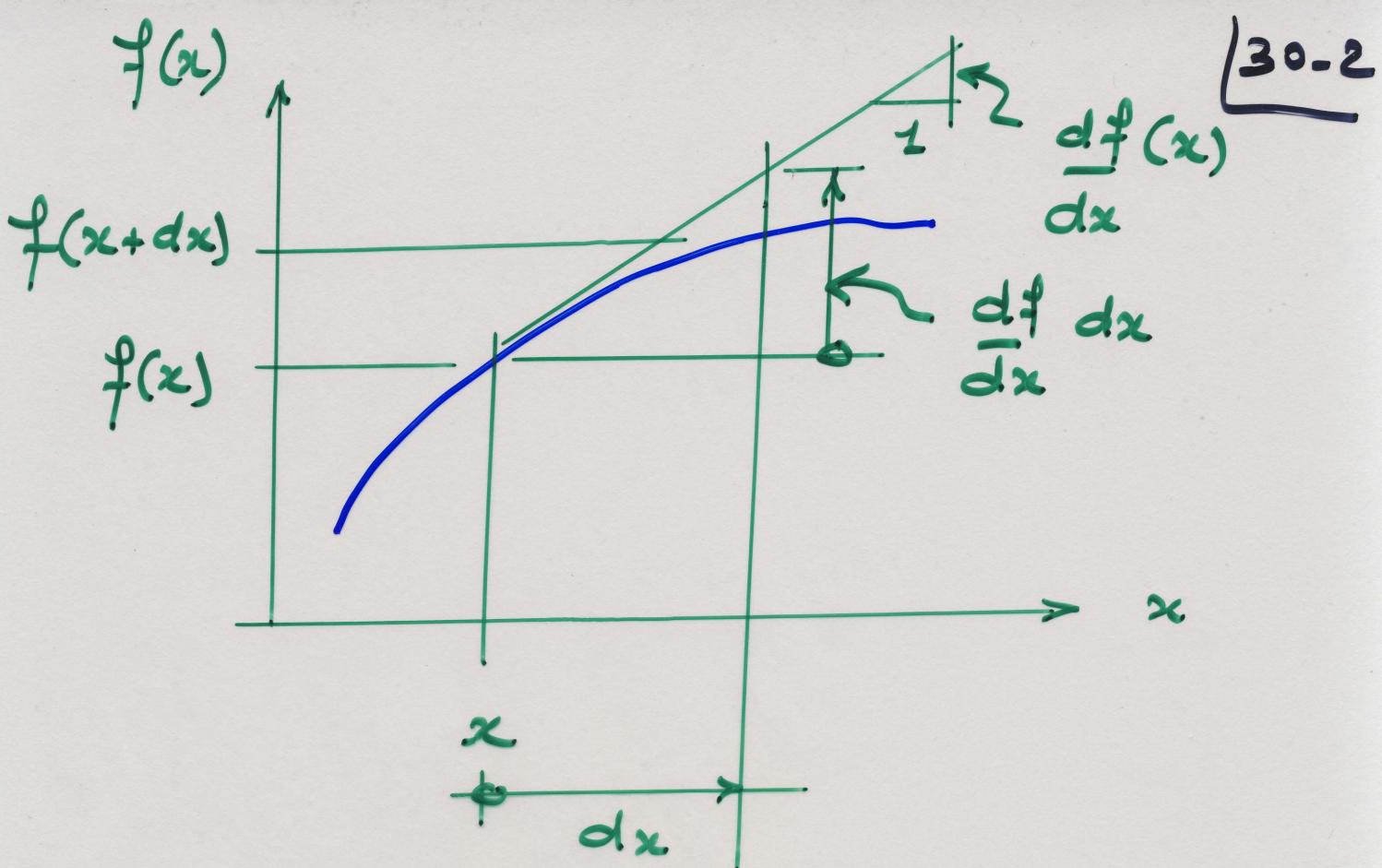
$$\begin{cases} \delta x_B = - (L \delta\theta) \cos \sin \theta \\ \delta y_B = (L \delta\theta) \cos \theta \end{cases}$$

Taylor Series expans. :

$$x_B(\theta) = L \cos \theta$$

$$x_B(\theta + \delta\theta) = L \cos(\theta + \delta\theta)$$

Recall:  $f(x + dx) = f(x) + \frac{df(x)}{dx} dx$   
 $+ h.o.t.$   
 higher order terms



$$x_B(\theta + \delta\theta) = x_B(\theta) + \frac{dx_B(\theta)}{d\theta} \delta\theta + h.o.t.$$

$$= x_B(\theta) + \delta x_B(\theta)$$

or  $\delta x_B(\theta) = \underbrace{x_B(\theta + \delta\theta) - x_B(\theta)}_{\frac{dx_B(\theta)}{d\theta} \delta\theta}$

Similarly for  $\delta y_B$ .

*x Comp. of  
Virt. disp.  
of B.*

# Beam Theory : Justification of 30-2

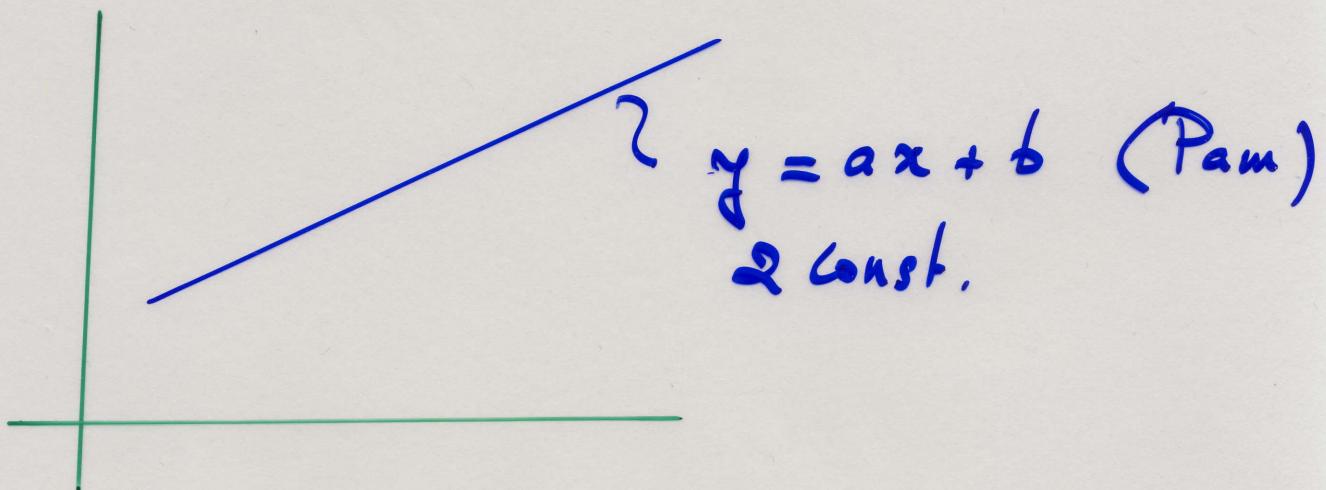
Sida : 2nd order diff eq.  $\frac{d^2y}{dx^2} = f(x)$  p.25-1

Gabriel : Defn of beam specified by poly.

Brian : soln to diff eq.

Euler-Bernoulli beam : 4th diff. eq.  
in terms of transv. disp.

Int.  $\Rightarrow$  4 const.



Peter : 4 const  $\Rightarrow$  3rd-order poly,

i.e.,  $c_0 + c_1 x + c_2 x^2 + c_3 x^3$

Mtg 31: Mon, 13 Nov 06 + 5 min  
(Lect 7)

31-1

Project (HW10, Part I) presentation.

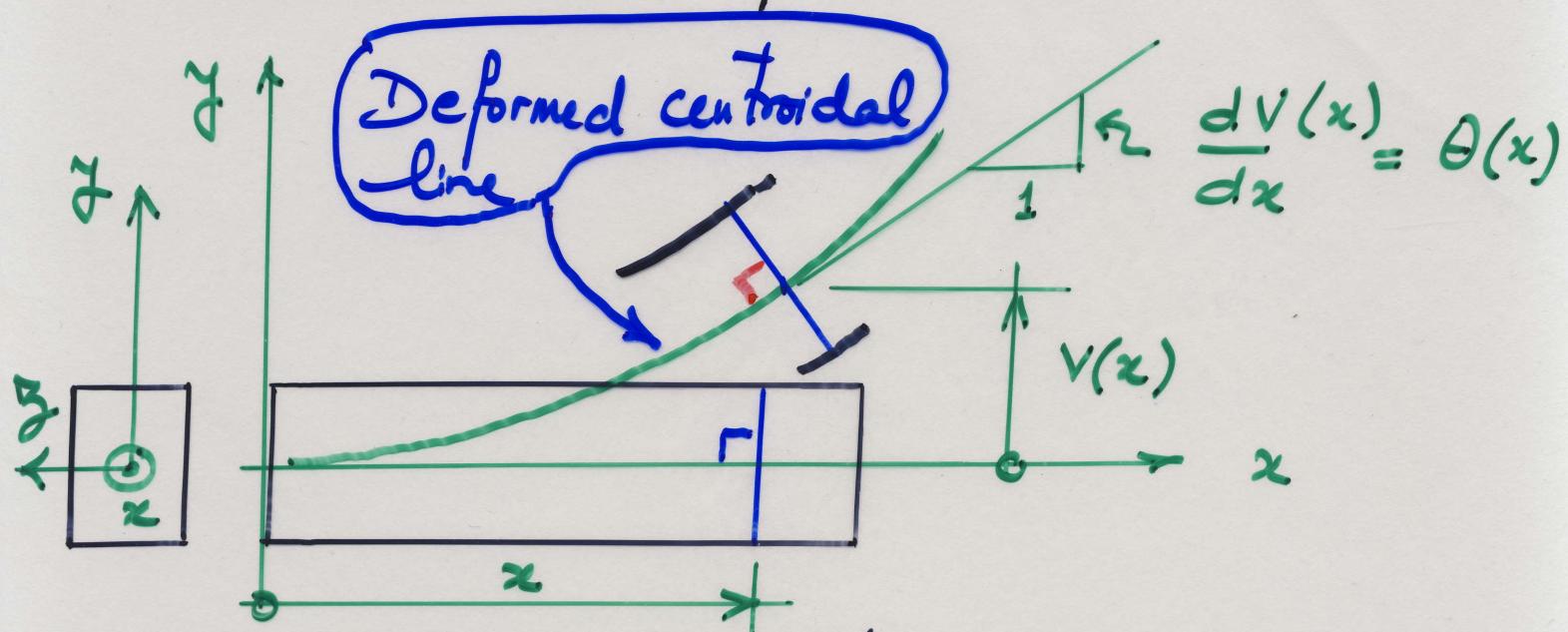
Mtg 32: Wed, 15 Nov 06 + 5 min  
(Lect 8)

32-1

## Beam theory (Cont'd)

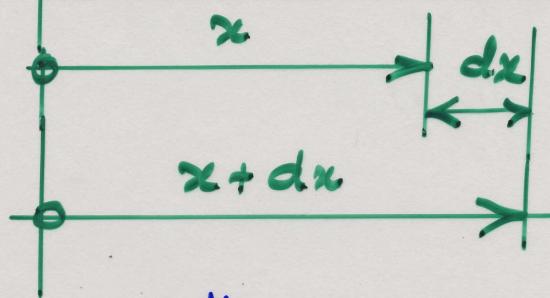
Euler-Bernoulli beam theory, fund.

assump.: Plane cross-section remains plane and perpendicular to deformed centroidal line of beam.



- \* Undeformed centroidal line coincides w/ x axis.
- \* Hooke's law:  $\sigma = E \epsilon$
- \* Q: How to relate  $\epsilon$  to  $v$  (since the beam diff. eq. is in terms of  $v$ ) ?

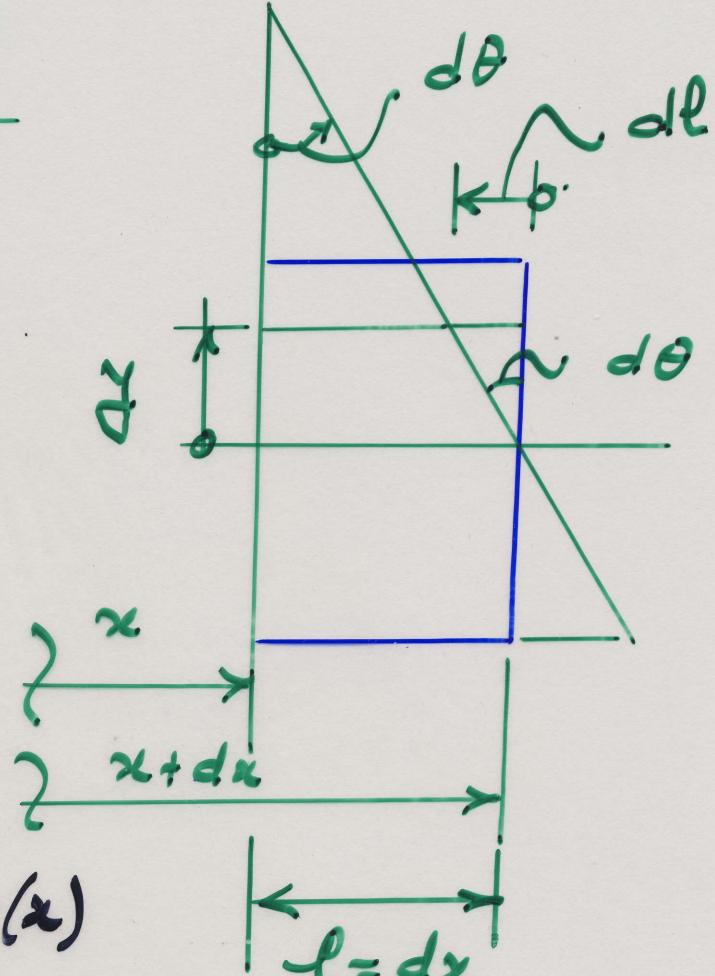
32-2



Chris : "FBD"

$d\ell = -\gamma d\theta$   
(shortening of  
fiber at ordinate  
 $y$ )

$$\varepsilon = \frac{d\ell}{\ell} = -\gamma \frac{d\theta(x)}{dx}$$



$\Rightarrow \varepsilon(x)$

$$\boxed{\varepsilon(x, y) = -\gamma \frac{d^2 v(x)}{dx^2}}$$

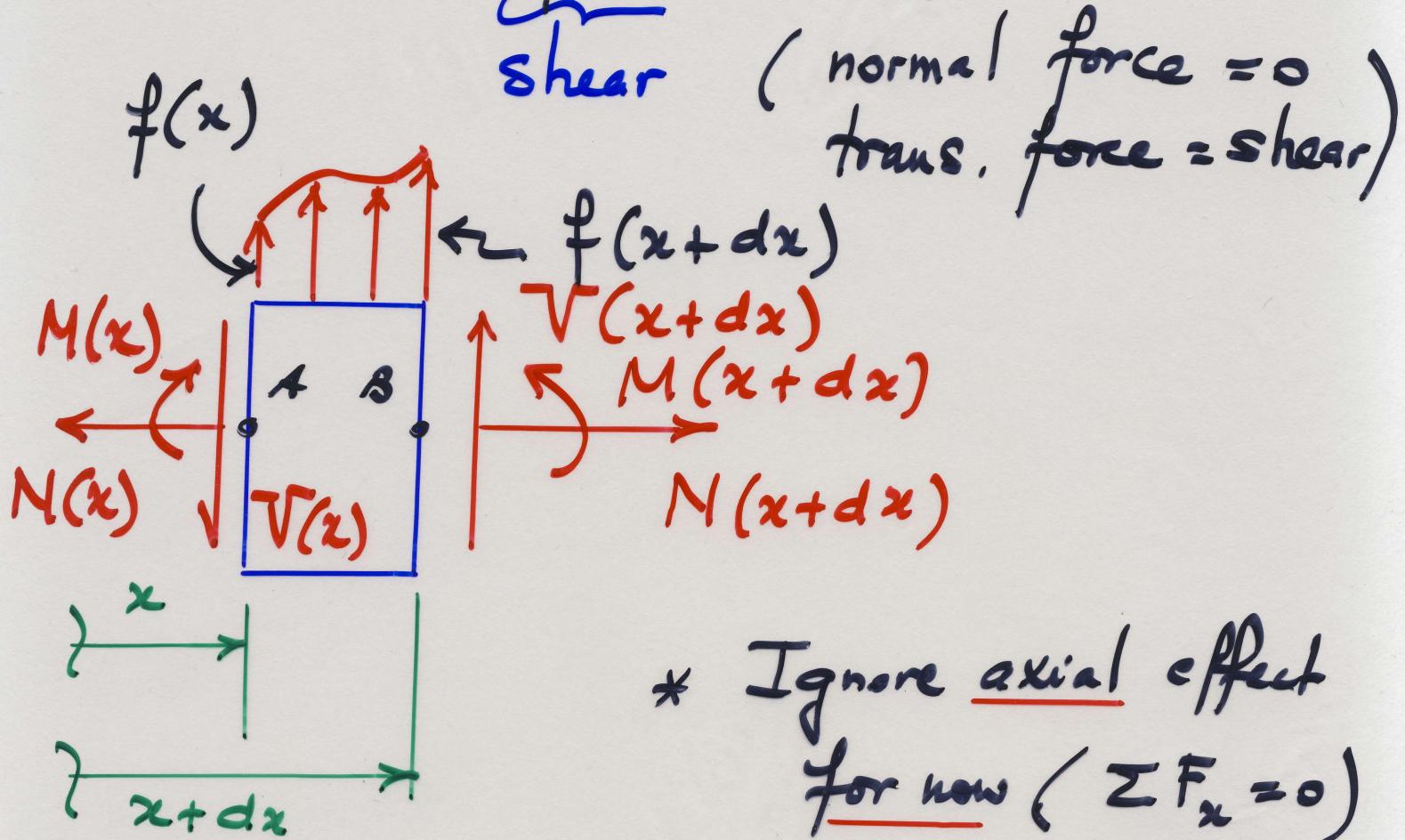
David: curvature

Hooke's law:

$$\boxed{\sigma(x, y) = -E \gamma v''(x)}$$

$$V''(x) := \frac{d^2 V(x)}{dx^2} \quad \text{curvature} \quad \underline{32-3}$$

Jarita : Compute the resultants  
(force, mom)



$$\begin{aligned} \sum F_y = 0 &= -V(x) + f(x) dx \\ &\quad + V(x+dx) \end{aligned}$$

David : Taylor Series

$$g(x+dx) = g(x) + \frac{dg(x)}{dx} dx + \underbrace{\text{h.o.t.}}_{\text{higher order terms}}$$

$$\Rightarrow \sum F_y = 0 = \frac{dV(x)}{dx} dx + \underbrace{h.o.t.}_{+ f(x) dx} \begin{matrix} (3^2 - 4 \\ dx^2, \\ dx^3, \dots) \end{matrix}$$

Ex:  $dx = 10^{-1}$  (0.1)

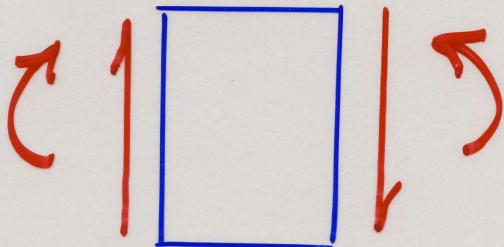
$$dx^2 = 10^{-2} \text{ (0.01)} < dx$$

Neglect hot : (±)

$$\frac{dV(x)}{dx} = -f(x)$$

HW: Find rel. betw.  $V$  and  $f$   
for the alternative convention :

Book p. 236



\* Can choose either A or B in Fig. on p. 32.3 to do mom. equil. (convenient since eliminate as many forces as possible).

Mtg 33: Thu, 16 Nov 06

33-1

Goal: Derive beam stiffness matrix on p. 25-1.

FBD on p. 32-3: Mom. equil. about pt A

$$+\sum M_A = 0 = -M(x) + f(x) dx \cdot \frac{dx}{2}$$

$$+ V(x+dx) dx + M(x+dx)$$

$$\xrightarrow{\text{Taylor series}} = \frac{dM(x)}{dx} dx + \cancel{hot} + f(x) \frac{dx^2}{2} + \left[ V(x) + \frac{dV(x)}{dx} dx + \cancel{hot} \right] dx$$

$$= \frac{dM(x)}{dx} dx + \cancel{hot} + V(x) dx$$

$$\begin{aligned} & \uparrow \\ & \text{neglect hot} + \left[ \frac{1}{2} f(x) + \frac{dV(x)}{dx} \right] dx^2 + \cancel{hot} \\ & (dx^2, dx^3, \dots) \end{aligned}$$

$$= \left[ \frac{dM(x)}{dx} + V(x) \right] dx$$

$\Rightarrow$

$$\frac{dM(x)}{dx} = -V(x)$$

(1)

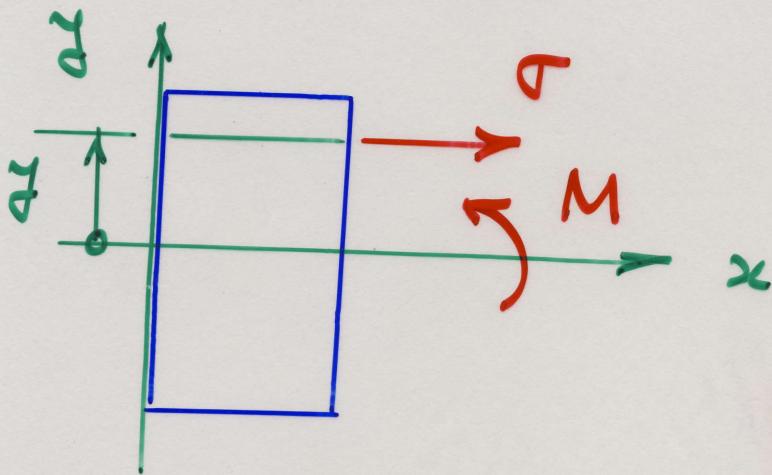
Book p. 237  
cliff conv.

Using Eq. (1) on p. 32-4 and  
 Eq. (1) on p. 33-1 : 33-2

$$\Rightarrow \frac{d^2M(x)}{dx^2} = - \frac{dV(x)}{dx} = -(-\ddot{\gamma}(x))$$

$$\Rightarrow \boxed{\frac{d^2M(x)}{dx^2} = \ddot{\gamma}(x)} \quad (1)$$

\* Relate  $M(x)$  to  $v(x)$  (transv. disp)



$$M = - \int (\sigma dA) z$$

*important*

$$M = - \int_A (-Ez v'' \underbrace{dA}) z$$

*dydz*

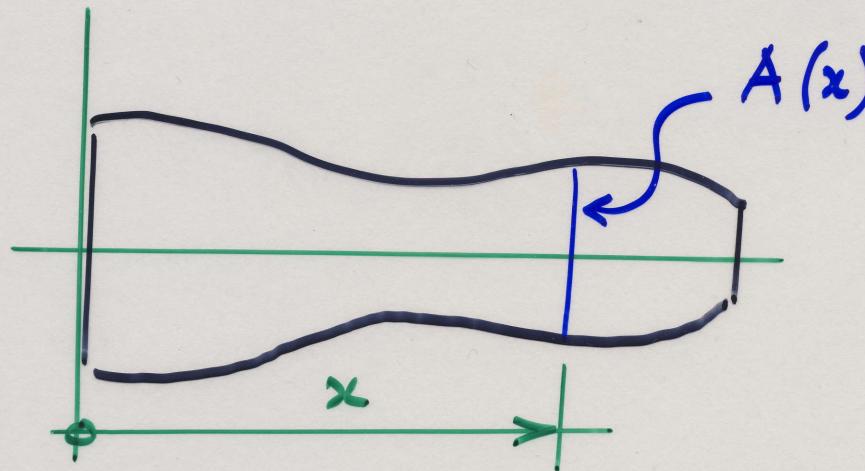
(see fig. of cross-section on p. 32-1)

$$M(x) = \int_A E z^2 v''(x) dy dz$$

$$M(x) = E \left( \underbrace{\int y^2 dA}_{I} \right) v''(x) \quad \boxed{33-3}$$

$$\Rightarrow M(x) = \underbrace{EI}_{\uparrow} \frac{d^2 v(x)}{dx^2} \quad (1)$$

Can be a func. of  $x$   
 for beams w/ variable  
 cross section along  $x$  axis, or  
 with  $E$   
 varying along  $x$  axis.



Eq. (1), p. 33-2 :

$$\frac{d^2}{dx^2} \left( EI(x) \frac{d^2 v(x)}{dx^2} \right) = f(x) \quad (2)$$

For the stiffness mat. on p. 25-1 33-4  
 (See Fig. on p. 24-4) :

$$\ddot{f}(x) = 0 \quad (\text{no dist. load})$$

$$EI(x) = \text{const.}$$

$$\Rightarrow EI \frac{d^4 v(x)}{dx^4} = 0$$

$$\Rightarrow \boxed{\frac{d^4 v(x)}{dx^4} = 0} \quad (1)$$

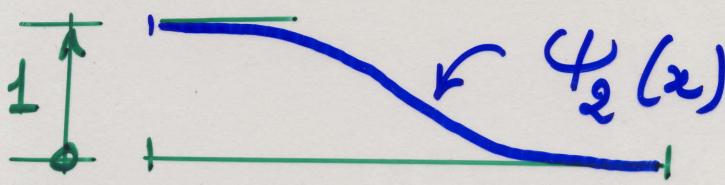
$$\Rightarrow \boxed{v(x) = C_0 + C_1 x + C_2 x^2 + C_3 x^3} \quad (2)$$

(see p. 30-3)

p. 25-1 :

$$\underline{k}_B^{(e)} \underline{d}_B^{(e)} = \underline{f}_B^{(e)} \quad (3)$$

$$\text{let } \underline{d}_B^{(e)} = \begin{Bmatrix} 1 \\ 0 \\ 0 \\ 0 \end{Bmatrix} \Rightarrow \underline{f}_B^{(e)} = \begin{Bmatrix} \underline{k}_{11}^{(e)} B \\ \vdots \\ \underline{k}_{41}^{(e)} B \end{Bmatrix} \quad (4)$$



1st col.

Mtg 34: Fri, 17 Nov 06 + 5 min 34-1  
 (Lect 9)

p. 33-4:  $\begin{cases} \tilde{d}_2^N = 1 & (\text{trans. disp @ node } \boxed{1}) \\ \tilde{d}_3^N = 0 & (\text{rot @ node } \boxed{1} \\ \Rightarrow \text{zero slope}) \end{cases}$

Similarly for node  $\boxed{2}$

$$\begin{cases} \tilde{d}_5^N = 0 & (\text{trans. disp.}) \\ \tilde{d}_6^N = 0 & (\text{rot. or slope}) \end{cases}$$

Comp.  $c_0, \dots, c_3$

$$\begin{cases} \tilde{d}_2^N = 1 = v(0) = c_0 \end{cases} \quad (1)$$

$$\begin{cases} \tilde{d}_5^N = 0 = v(L) = 1 + c_1 L + c_2 L^2 + c_3 L^3 \end{cases} \quad (2)$$

$$v'(x) = c_1 + 2c_2 x + 3c_3 x^2$$

$$\begin{cases} \tilde{d}_3^N = 0 = v'(0) = c_1 \end{cases} \quad (3)$$

$$\begin{cases} \tilde{d}_6^N = 0 = v'(L) = 2c_2 L + 3c_3 L^2 \end{cases}$$

$$\Rightarrow c_2 = -\frac{3}{2} c_3 L \quad (4)$$

Using (3), (4) in (2) p. 34-1 : 34-2

$$0 = 1 + 0 + C_3 \left( \underbrace{-\frac{3}{2} L^3 + L^3}_{-\frac{L^3}{2}} \right)$$

$\Rightarrow$

$$C_3 = \frac{2}{L^3}$$

$$C_2 = -\frac{3}{L^2}$$

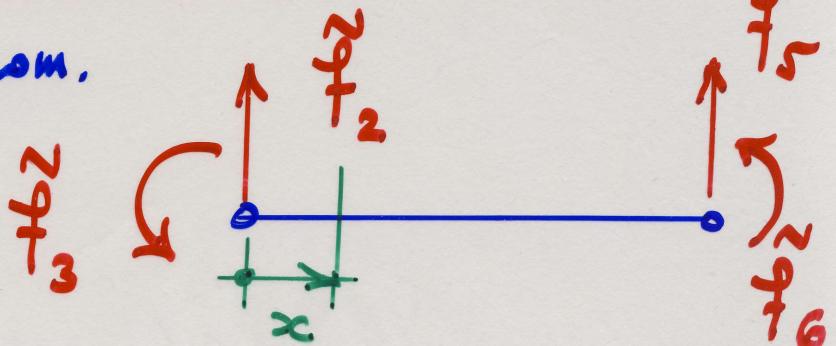
$$V(x) = 1 - \frac{3}{L^2} x^2 + \frac{2}{L^3} x^3$$

$$V''(x) = 2C_2 + 6C_3 x$$

$$M(x) = EI V''(x)$$

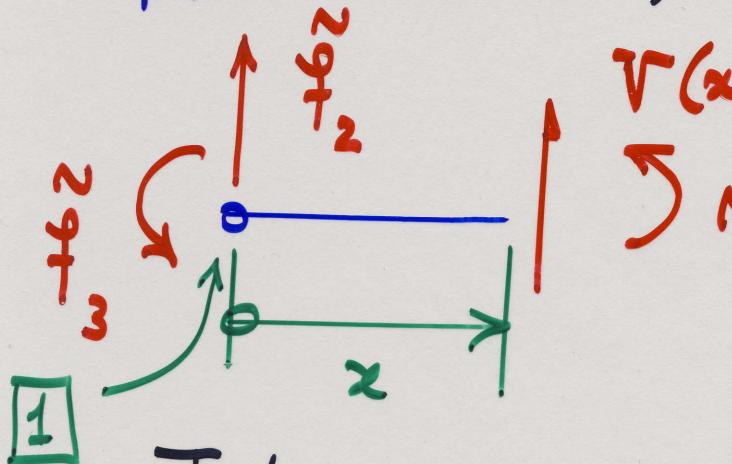
$$= EI \left[ -\frac{C_2}{L^2} + \frac{12}{L^3} x \right]$$

internal mom.



Pam:  $x = 0$ ,  $x = L$

34-3



Jarita:

$$\Rightarrow \tilde{f}_2 = -\tilde{V}(0)$$

Take  $x \rightarrow 0$ , do equil. of mom:  
about 1:

$$+\sum M_{\boxed{1}} = 0 = \tilde{f}_3 + M(0)$$

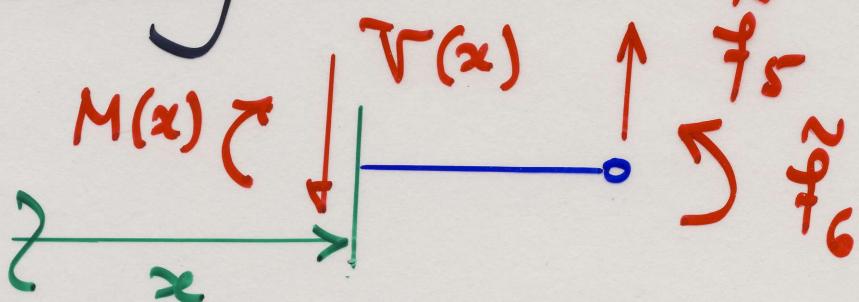
$$\Rightarrow \tilde{f}_3 = -M(0) = +\frac{6EI}{L^2}$$

$$= \tilde{k}_{21} = \tilde{k}_{32}$$

See  $\tilde{k}_B$

on p. 25-1

Similarly at  $x = L$ :



$$\Rightarrow \tilde{f}_5 = +V(L)$$

David:

Take  $x \rightarrow L$ ,

$$\tilde{f}_5 = M(L)$$

$$\tilde{k}_{41} = \tilde{k}_{62} = \frac{6EI}{L^2}$$

Using (1) of p. 33-1 and

(34-4)

(1) of p. 33-3 :

$$T(x) = - \frac{dM}{dx} = - EI \frac{d^3 v(x)}{dx^3}$$

$$T(x) = - 12 \frac{EI}{L^3} \quad \text{for all } x.$$

$$\Rightarrow \tilde{\phi}_2 = 12 \frac{EI}{L^3} = - \tilde{\phi}_5$$

Mtg 35: Mon, 20 Nov 06 + 5 min 135-1  
(lecture)

Goal: Explain 3rd col (also 3rd  
row) corrsp.  $\tilde{d}_3$  in  $\underline{k}^{(e)}$   $6 \times 6$ , p.

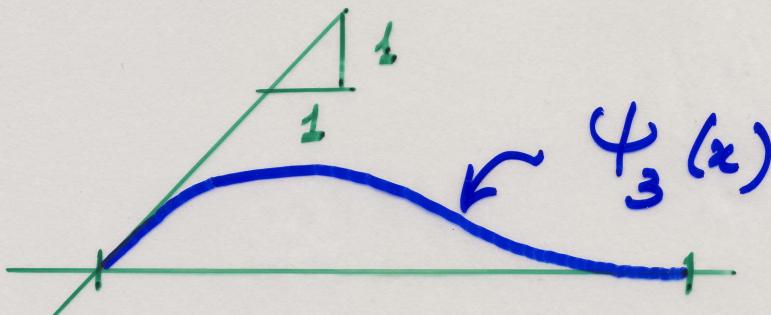
Method: (see p. 33-4) 25-3.

Select

$$\underline{\tilde{d}}^{(e)} = \begin{Bmatrix} 0 & 0 & 1 \\ 0 & 0 & 0 \end{Bmatrix}$$

$\tilde{d}_3$

Eqs (1), (2), p. 33-4.



$$V(x) = c_0 + c_1 x + c_2 x^2 + c_3 x^3$$

$$V'(x) = c_1 + 2c_2 x + 3c_3 x^2$$

$$V''(x) = 2c_2 + 6c_3 x$$

35-2

$$v'''(x) = 6c_3$$

$$v(0) = \tilde{d}_2 = 0 \quad | \quad v(L) = \tilde{d}_5 = 0$$

$$v'(0) = \tilde{d}_3 = 1 \quad | \quad v'(L) = \tilde{d}_6 = 0$$

Larry:

$$\begin{aligned} c_0 &= 0 & c_2 &= -\frac{2}{L} \\ c_1 &= 1 & c_3 &= \frac{1}{L^2} \end{aligned}$$

$$v(x) = \underbrace{\frac{1}{2} \frac{x^2}{L}}_{\frac{x}{2}} - \frac{2}{L} x^2 + \frac{1}{L^2} x^3$$

Tony:  $M(x) = EI v''(x)$

$$= EI \left[ -\frac{4}{L} + \frac{6}{L^2} x \right]$$

Sida:  $V(x) = -\underline{\frac{dM(x)}{dx}} = -EI v'''(x)$

$$= -\underline{\frac{6EI}{L^2}}$$

Jarita: FBD's, p. 34-3

35-3

At node  $\boxed{1}$ , as  $x \rightarrow 0$ :

Shear:  $\tilde{f}_2 = -V(0) = +\frac{6EI}{L^2} = \tilde{k}_{23}$

Mom:  $\tilde{f}_3 = -M(0) = \frac{4EI}{L} = \tilde{k}_{33}$

At node  $\boxed{2}$ , as  $x \rightarrow L$ :

Shear:  $\tilde{f}_5 = +V(L) = -\frac{6EI}{L^2} = \tilde{k}_{53}$

Mom:  $\tilde{f}_6 = +M(L) = \frac{2EI}{L} = \tilde{k}_{63}$

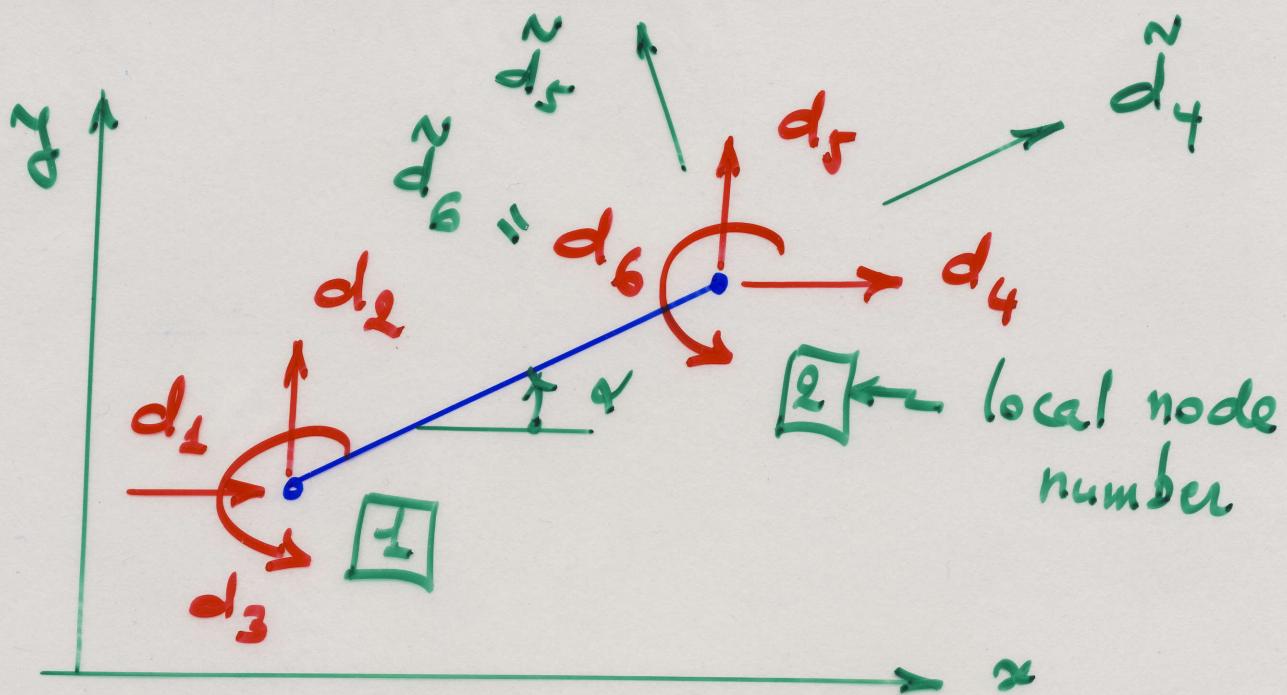
HW: Derive all coeff in 5th col.  
and 6th col. in  $\underline{\tilde{k}}^{(e)}_{6 \times 6}$ .

Mtg 36 : Mon, 27 Nov 06,

36-1

- Frame elem arb. oriented in space (2-D)
- Distr. load (transv.) on frame elem (HW10).

General 2-D frame elem:



In local coord :  $\underline{\underline{k}}^{(e)} \underline{\underline{d}}^{(e)} = \underline{\underline{f}}^{(e)}$

$6 \times 6 \quad 6 \times 1 \quad 6 \times 1$

(P. 25-3a)

P. 24-4 : dof's in local coord.  $\underline{\underline{d}}^{(e)}$

$6 \times 1$

$$\begin{matrix} \text{Chris :} \\ \text{Shawn :} \end{matrix} \quad \frac{\underline{d}^{(e)}}{6 \times 1} = \frac{\underline{T}^{(e)}}{6 \times 6} \underline{d}^{(e)} \quad (36-2)$$

$$\frac{\underline{T}^{(e)}}{6 \times 6} = \left[ \begin{array}{cc} \underline{R}^{(e)}_{3 \times 3} & \underline{0}_{3 \times 3} \\ \underline{0}_{3 \times 3} & \underline{R}^{(e)}_{3 \times 3} \end{array} \right] \quad 6 \times 6$$

Refer

$$\underline{R}^{(e)}_{3 \times 3} = \left[ \begin{array}{cc|c} l^{(e)} & m^{(e)} & 0 \\ -m^{(e)} & l^{(e)} & 0 \\ \hline 0 & 0 & 1 \end{array} \right] \quad 3 \times 3$$

(See p. 8-2,  
8-3)

$\underline{R}^{(e)}$  is orthog. mat.  $\Rightarrow$

$$\underline{T}^{(e)-1} = \left[ \begin{array}{cc} \underline{R}^{(e)-1} & \underline{0} \\ \underline{0} & \underline{R}^{(e)-1} \end{array} \right] \quad \underline{R}^{(e)-1} = \underline{R}^{(e)T} \quad (\text{p. 9-2})$$

$$\Rightarrow \underline{\underline{T}}^{(e)-1} = \begin{bmatrix} \underline{\underline{R}}^{(e)T} & \underline{\underline{0}} \\ \underline{\underline{0}} & \underline{\underline{R}}^{(e)T} \end{bmatrix} \stackrel{36-3}{=} \underline{\underline{T}}^{(e)T}$$

Result :

$$\boxed{\left[ \underline{\underline{T}}^{(e)T} \underline{\underline{k}}^{(e)} \underline{\underline{T}}^{(e)} \right] \underline{\underline{d}}^{(e)} = \underline{\underline{T}}^{(e)T} \underline{\underline{\varphi}}^{(e)}}$$

$\underline{\underline{k}}^{(e)}$        $\underline{\underline{\varphi}}^{(e)}$

obtained by 2 methods (similar to truss elem.) :

1) By transf. of coord.

$$\underline{\underline{k}}^{(e)} \underline{\underline{d}}^{(e)} = \underline{\underline{\varphi}}^{(e)}$$

$$\underline{\underline{k}}^{(e)} \underline{\underline{T}}^{(e)} \underline{\underline{d}}^{(e)} = \underline{\underline{T}}^{(e)} \underline{\underline{\varphi}}^{(e)}$$

since  $\underline{\underline{\varphi}}^{(e)} = \underline{\underline{T}}^{(e)} \underline{\underline{\varphi}}^{(e)}$

$$\text{Same as } \underline{\tilde{d}}^{(e)} = \underline{T}^{(e)} \underline{\tilde{d}}^{(e)} \quad (36-4)$$

$$\Rightarrow \underbrace{\underline{T}^{(e)-1}}_{\underline{T}^{(e)T}} \underbrace{\underline{k}^{(e)}}_{\underline{k}^{(e)}} \underline{T}^{(e)} \underline{\tilde{d}}^{(e)} = \underline{T}^{(e)-1} \underline{\tilde{f}}^{(e)}$$

$$\underline{T}^{(e)} \underline{\tilde{f}}^{(e)}$$

$$\text{rhs} = \underline{\tilde{f}}^{(e)}$$

2) Prince. Virt. Work :

$$\underline{\tilde{k}}^{(e)} \underline{\tilde{d}}^{(e)} = \underline{\tilde{f}}^{(e)}$$

$$PVW : \underline{\alpha}^{(e)} \cdot \left[ \underline{\tilde{k}}^{(e)} \underline{\tilde{d}}^{(e)} - \underline{\tilde{f}}^{(e)} \right] = 0$$

for all  $\underline{\alpha}^{(e)}$

$$\text{Transf. } \underline{\tilde{d}}^{(e)} = \underline{T}^{(e)} \underline{\tilde{d}}^{(e)}$$

$$\underline{\alpha}^{(e)} = \underline{T}^{(e)} \underline{\tilde{c}}^{(e)}$$

$$\text{or } \underline{\tilde{c}}^{(e)} = \underline{T}^{(e)} \underline{\tilde{c}}^{(e)}$$

$\underline{\underline{c}}^{(e)} = \underline{\underline{\alpha}}^{(e)} = \text{virt. disp.}$       36-5  
 in local coord.

$\underline{\underline{c}}^{(e)} = \text{virt. disp. in global coord.}$

$$\Rightarrow [\underline{\underline{T}}^{(e)} \underline{\underline{c}}^{(e)}] \cdot \left[ \begin{array}{c} \underline{\underline{k}}^{(e)} - \underline{\underline{T}}^{(e)} \underline{\underline{d}}^{(e)} \\ - \underline{\underline{\tilde{f}}}^{(e)} \end{array} \right] = 0$$

for all  $\underline{\underline{c}}^{(e)}$

$$\Rightarrow \underline{\underline{c}}^{(e)} \cdot \left[ \begin{array}{c} \underline{\underline{T}}^{(e)T} \underline{\underline{k}}^{(e)} \underline{\underline{T}}^{(e)} \underline{\underline{d}}^{(e)} \\ - \underline{\underline{T}}^{(e)T} \underline{\underline{\tilde{f}}}^{(e)} \end{array} \right] = 0$$

for all  $\underline{\underline{c}}^{(e)}$

$$\Rightarrow \underline{\underline{T}}^{(e)T} \underline{\underline{k}}^{(e)} \underline{\underline{T}}^{(e)} \underline{\underline{d}}^{(e)} = \underline{\underline{T}}^{(e)T} \underline{\underline{\tilde{f}}}^{(e)}$$

See mfg 21.